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# TRI-DIRECTIONAL LOADING TESTS ON REINFORCED CONCRETE SHEAR WALLS

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# TRI-DIRECTIONAL LOADING TESTS ON REINFORCED CONCRETE SHEAR WALLS

## Contents

1. Introduction
2. Outline of experiment
3. Experimental results
4. Conclusion

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Special Project for Reducing Vulnerability for Urban Mega Earthquake Disasters  
(ii) Maintenance and Recovery of Functionality in Urban Infrastructures



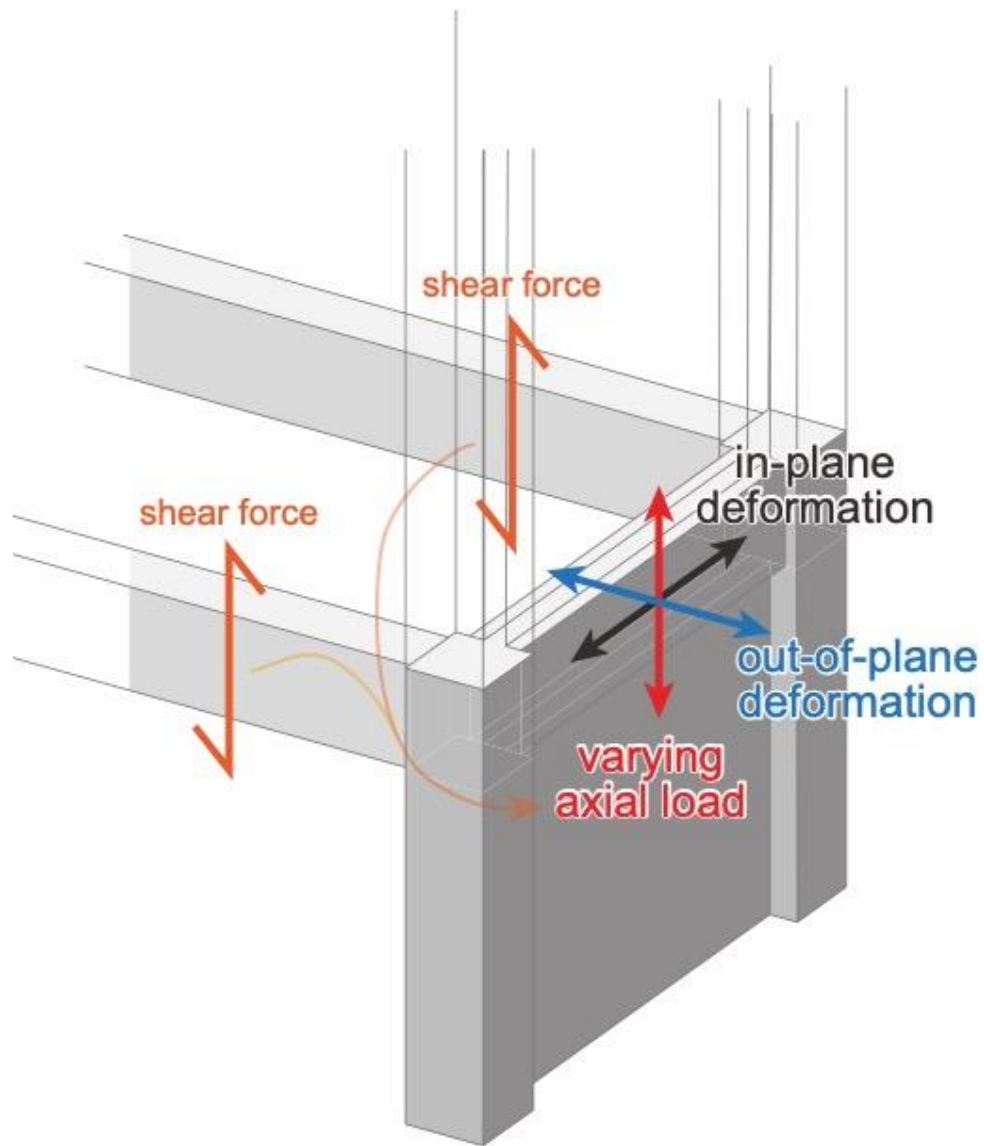
Special Project for Reducing Vulnerability for Urban Mega Earthquake Disasters  
(ii) Maintenance and Recovery of Functionality in Urban Infrastructures

#3-5(JMA Kobe 140%-1) 1/2



#3-5(JMA Kobe 140%-1) 1/2  
wall in X-2 frame on the 2nd floor





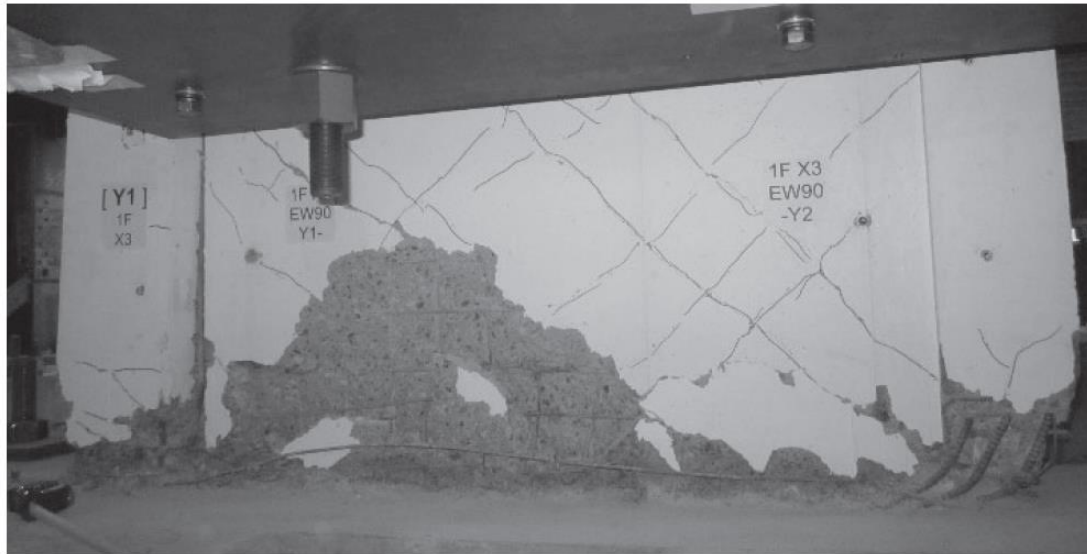
# Introduction

## Background

- At present, structural performance evaluation method of RC members considering tri-directional forces in earthquakes has yet to be established.

$$Q_w = \left\{ \frac{0.068 p_{te}^{0.23} (F_c + 18)}{\sqrt{\frac{M}{QD} + 0.12}} + 0.85 \sqrt{p_{wh} \sigma_{wh}} + 0.1 \sigma_o \right\} t_e j$$

$$\sigma_o = \frac{N}{A_g} \quad \begin{array}{l} \text{axial load} \\ \text{total area} \end{array}$$



Shaking table tests  
in E-Defense



# Introduction

## Objectives

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- investigating effects of **axial load condition** and **lateral bi-directional loading path** on the shear capacity of shear walls

## Methodology

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- conducting **loading tests on shear wall specimens subjected to bi-directional loading under various axial load**

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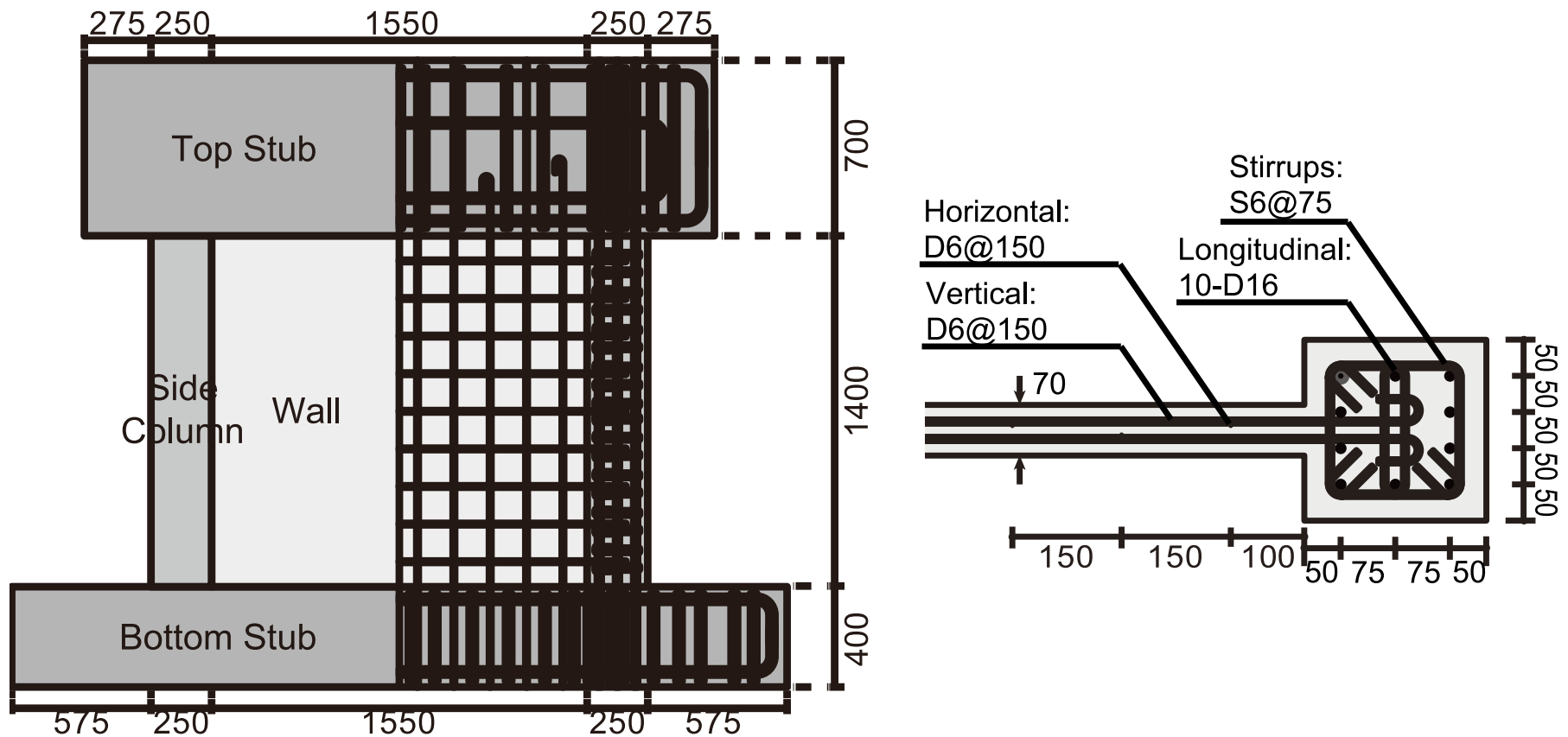
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# Outline of Loading Tests

## Outline of specimen

- Specimens: 30%-scale shear walls with identical dimensions and reinforcement arrangements.



# Outline of Loading Tests

## Experimental parameters

- Parameters: Axial load condition, O/I drift ratio  
(O/I drift ratio = out-of-plane drift angle / in-plane drift angle)

O/I drift ratio \ Axial load	constant		varying	
	$0.12A_g f'_c$	$0.20A_g f'_c$	0 $\sim 0.20A_g f'_c$	$-0.33a_{cg} f_y$ $\sim 0.20A_g f'_c$
0	WB00-C12 <sup>1)</sup>			
1.5	WB15-C12 <sup>1)</sup>	WB15-C20	WB15-C20T00	WB15-C20T33
3.0	WB30-C12 <sup>1)</sup>		WB30-C20T00	

$A_g$  gross sectional area of concrete  
 $f'_c$  compressive strength of concrete  
 $a_{cg}$  gross sectional area of long. rebars  
 $f_y$  yield strength of long. rebars

# Outline of Loading Tests

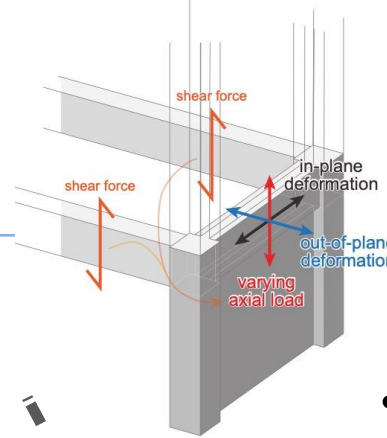
## Material

- There was a difference in the compressive strength of concrete.

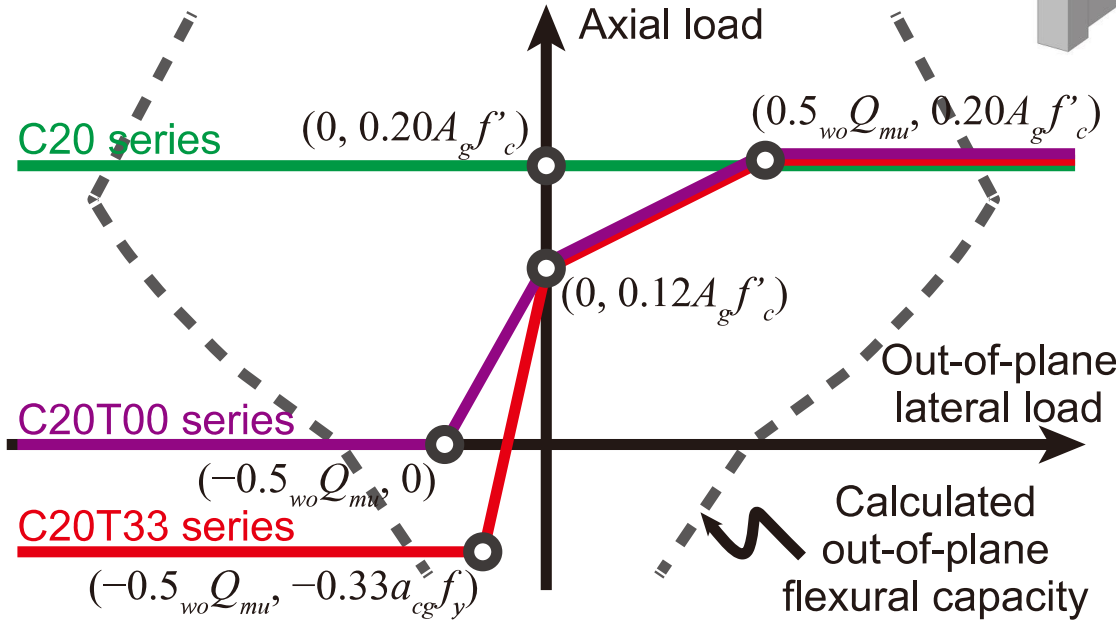
O/I drift ratio	Axial load	constant		varying	
		$0.12A_g f'_c$	$0.20A_g f'_c$	0 $\sim 0.20A_g f'_c$	$-0.33a_{cg} f_y$ $\sim 0.20A_g f'_c$
0		28.9			
1.5		28.9	21.9	23.0	26.9
3.0		28.9		24.5	

unit: MPa

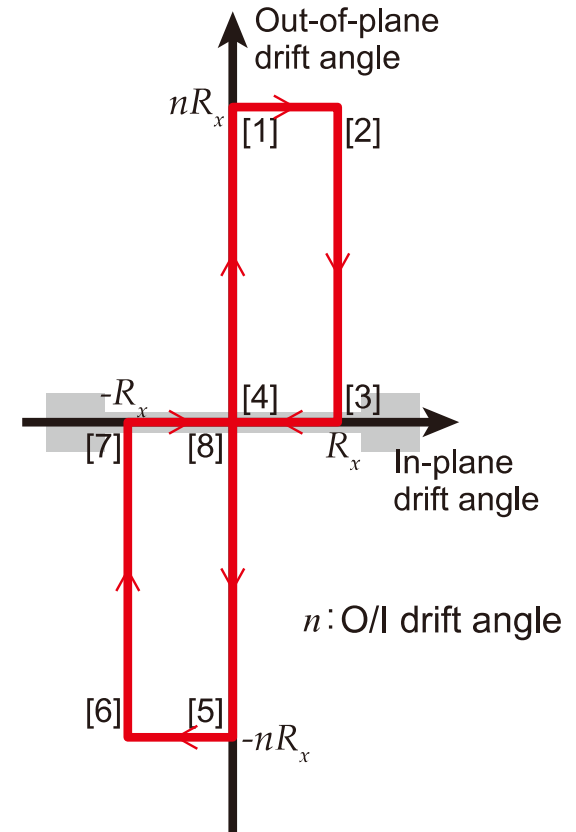
# Loading protocol



- Axial load



- Lateral loading path



- Out-of-plane flexural capacity is calculated as two columns ignoring the effect of wall panel.
- $R_x = 0.05, 0.10, 0.25, 0.50, 0.75, 1.0, 2.0\%$

# Outline of Loading Tests

## Calculation of capacity

- Flexural and shear ultimate capacity under the maximum axial load was calculated according to the equations of practical design in Japan.

O/l drift ratio	Axial load	constant		varying
		$0.20A_g f'_c$	0 $\sim 0.20A_g f'_c$	$-0.33a_{cg} f_y$ $\sim 0.20A_g f'_c$
1.5		WB15-C20	WB15-C20T00	WB15-C20T33
3.0			WB30-C20T00	

# Outline of Loading Tests

## Capacity calculation – in-plane

- It is estimated that **shear failure** precedes in the in-plane direction.

O/ drift ratio	Axial load	constant	0	varying
		$0.20A_g f'_c$	$\sim 0.20A_g f'_c$	$-0.33a_{cg} f_y$ $\sim 0.20A_g f'_c$
1.5		676 ▲ 1422	692 ▲ 1449	750 ▲ 1546
	3.0		714 ▲ 1486	In-plane shear flexural unit: kN



# Outline of Loading Tests

## Capacity calculation – out-of-plane

- It is estimated that **flexural failure** precedes in the out-of-plane direction.

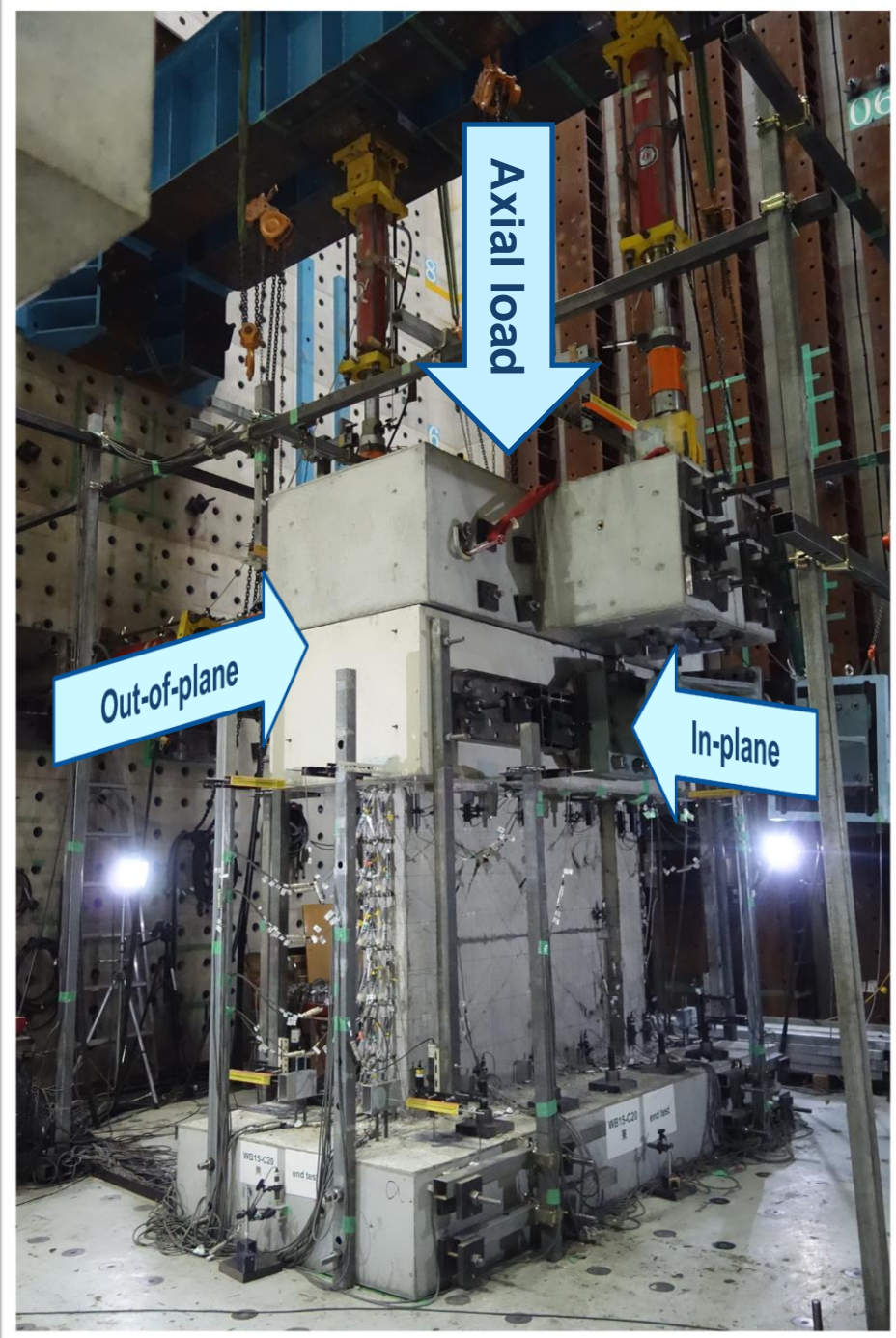
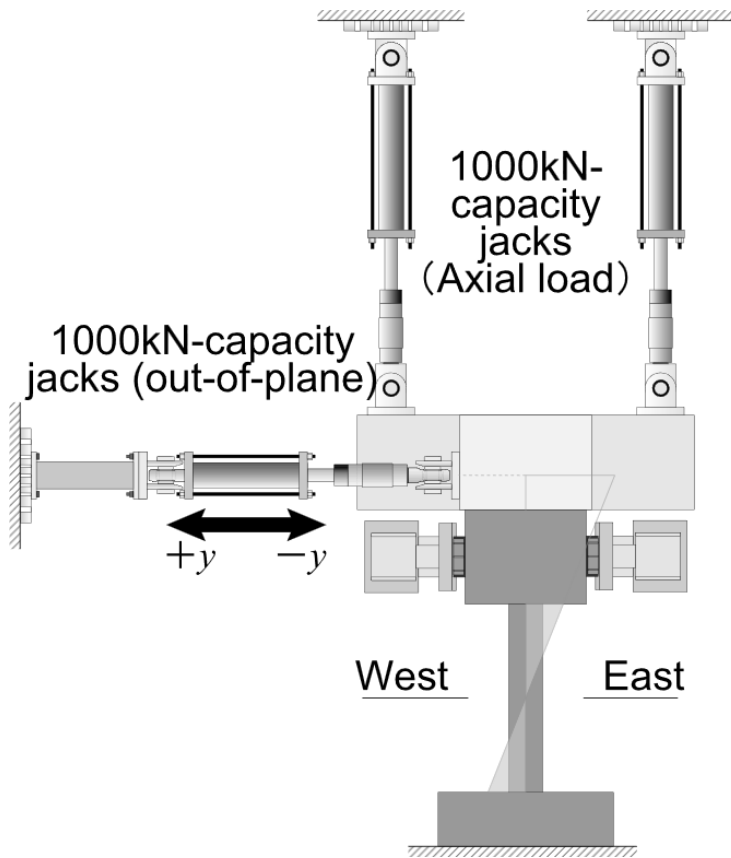
O/I drift ratio	Axial load	constant	0	varying
		$0.20A_g f'_c$	$\sim 0.20A_g f'_c$	$-0.33a_{cg} f_y$ $\sim 0.20A_g f'_c$
1.5		371 ▼ 215	377 ▼ 220	398 ▼ 238
	3.0		385 ▼ 227	Out-of-plane shear flexural

unit: kN

# Outline of L

## Loading setup

- In-plane: **cantilever** (shear span ratio)
- Out-of-plane: **double-curvature** (shear span ratio)





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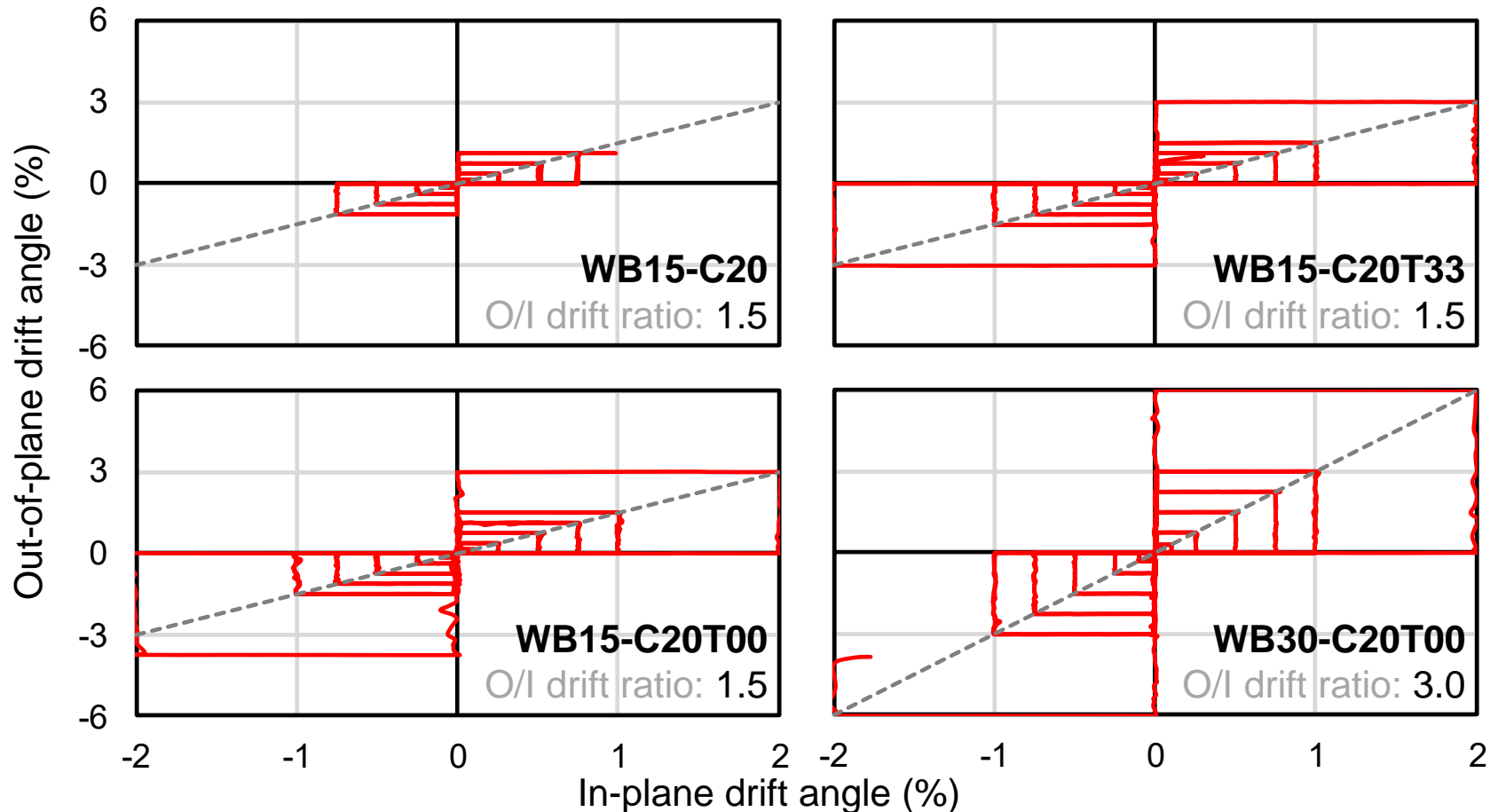
1. Introduction
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# Experimental Results

## Loading path

- The loading was carried out as planned generally.

## Lateral loading path

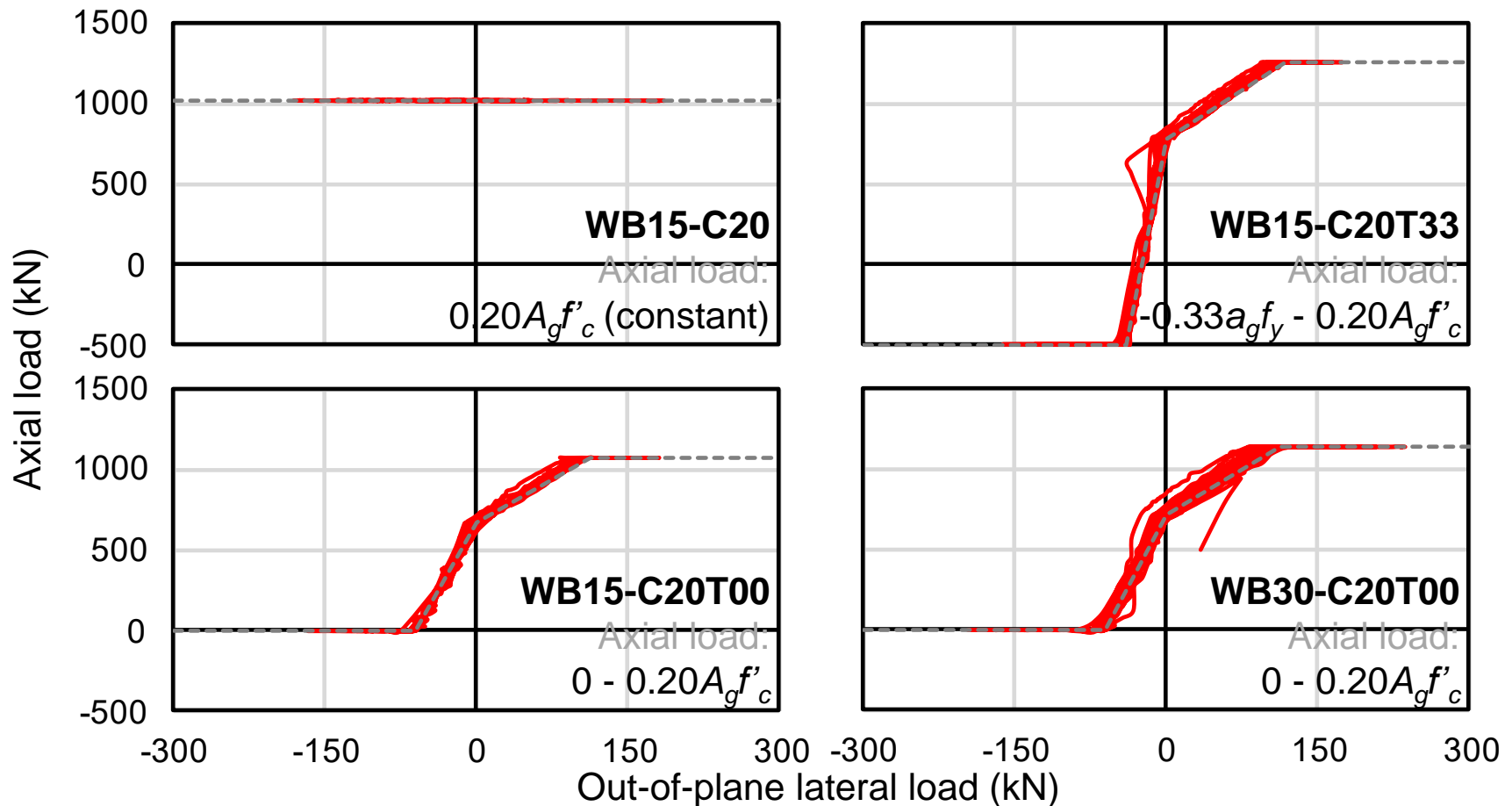


# Experimental Results

## Loading path

- The loading was carried out as planned generally.

## Axial loading path

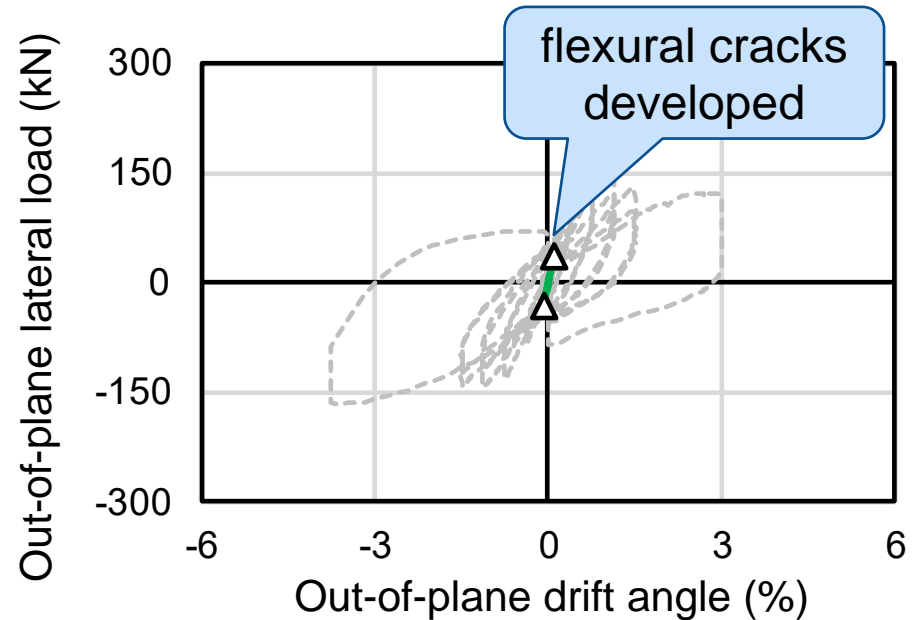
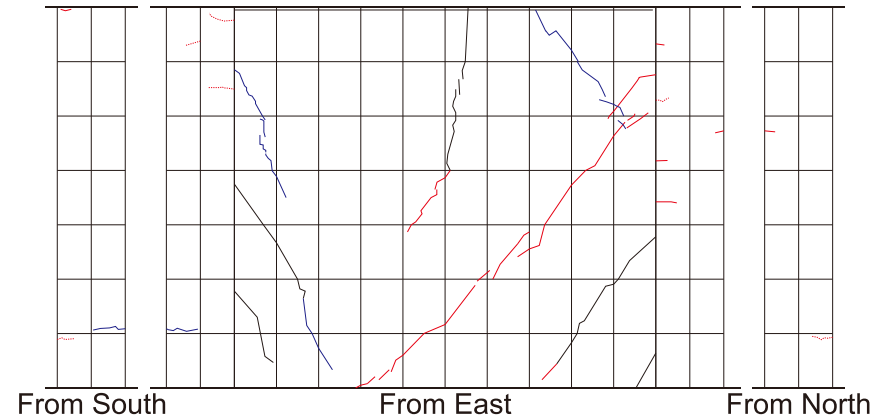
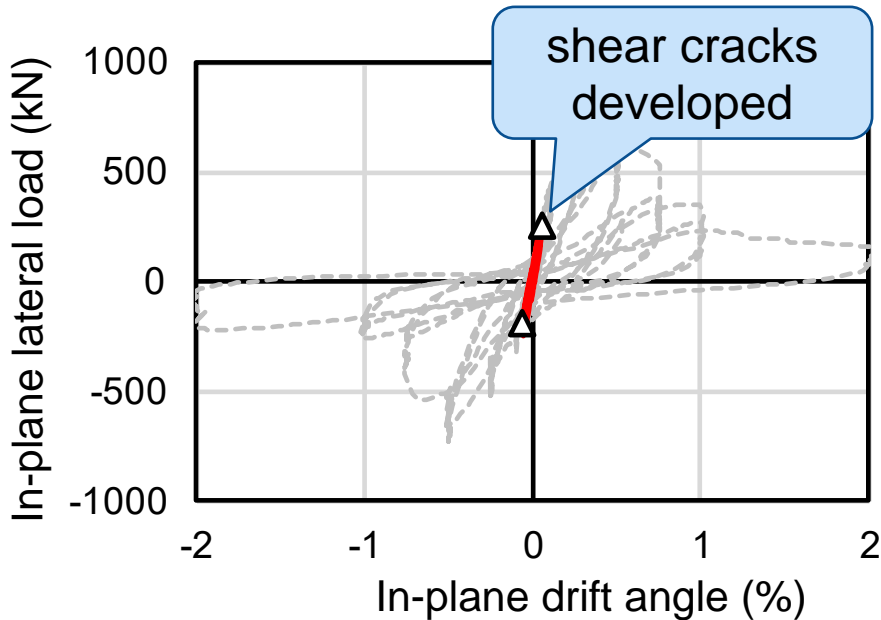


# Experimental Results

WB15-C20T00 O/l drift ratio: 1.5, Axial load:  $0 \sim 0.20A_g f'_c$

until  $R_x=0.05\%$

- Shear cracks were observed on the wall panel.
- Flexural cracks were observed on the side columns.

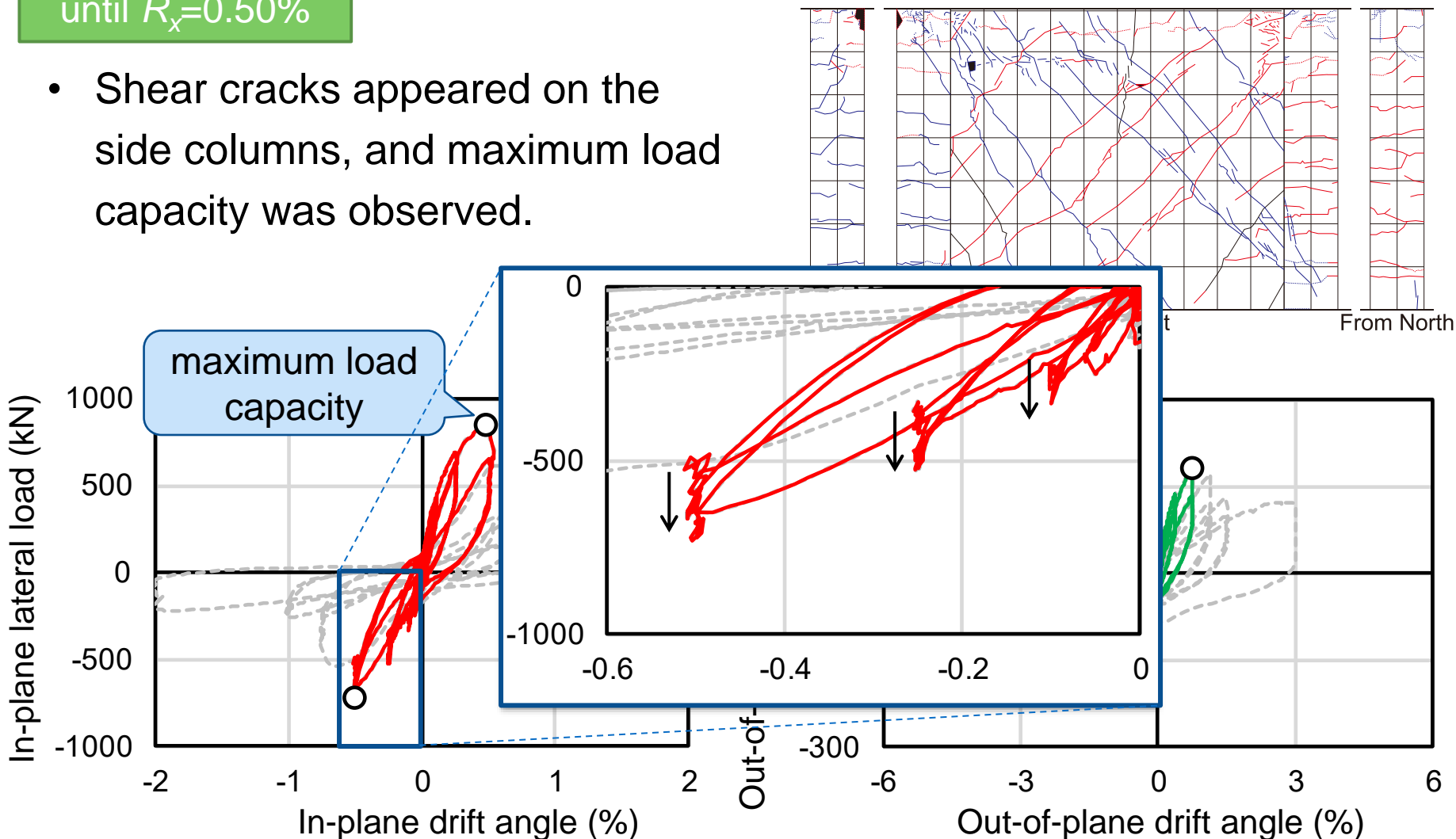


# Experimental Results

WB15-C20T00 O/l drift ratio: 1.5, Axial load:  $0 \sim 0.20A_g f'_c$

until  $R_x=0.50\%$

- Shear cracks appeared on the side columns, and maximum load capacity was observed.



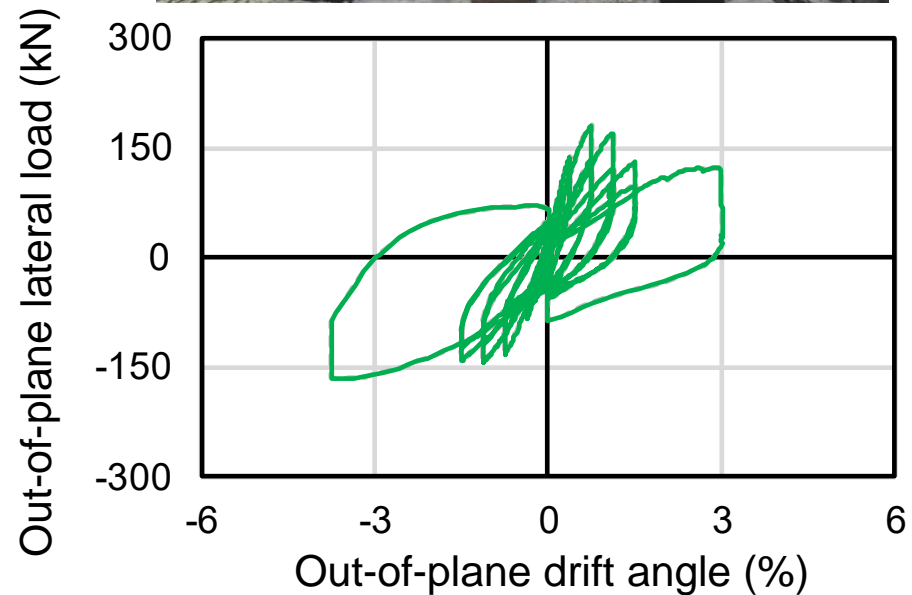
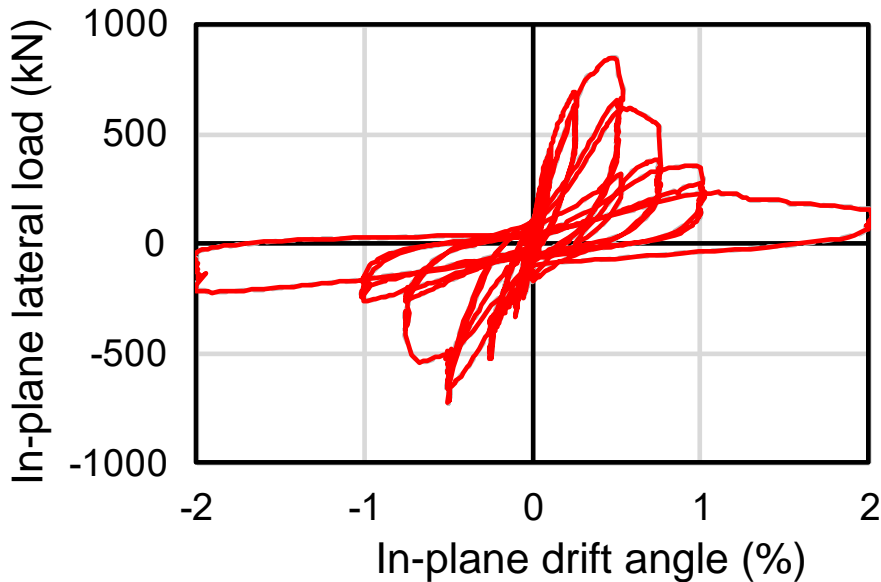
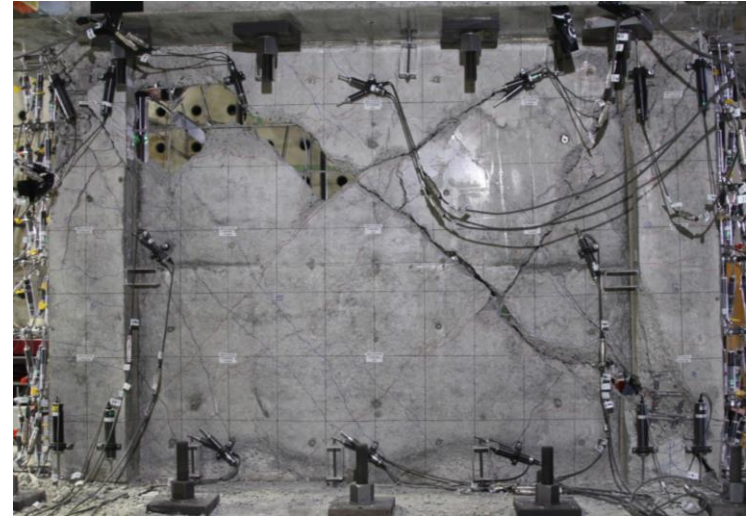


# Experimental Results

WB15-C20T00 O/l drift ratio: 1.5, Axial load:  $0 \sim 0.20A_g f'_c$

after test

- The concrete of the wall panel spalled, and the lateral load decreased.
- The loading was ceased after the completion of 1<sup>st</sup> cycle for  $R_x=2.0\%$ .

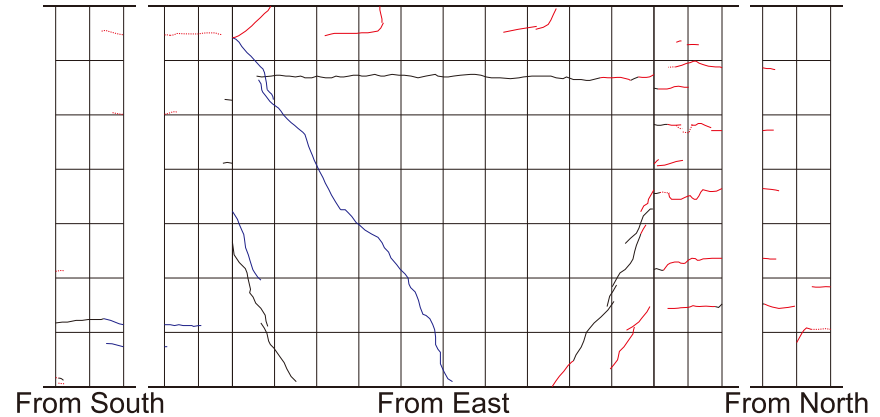
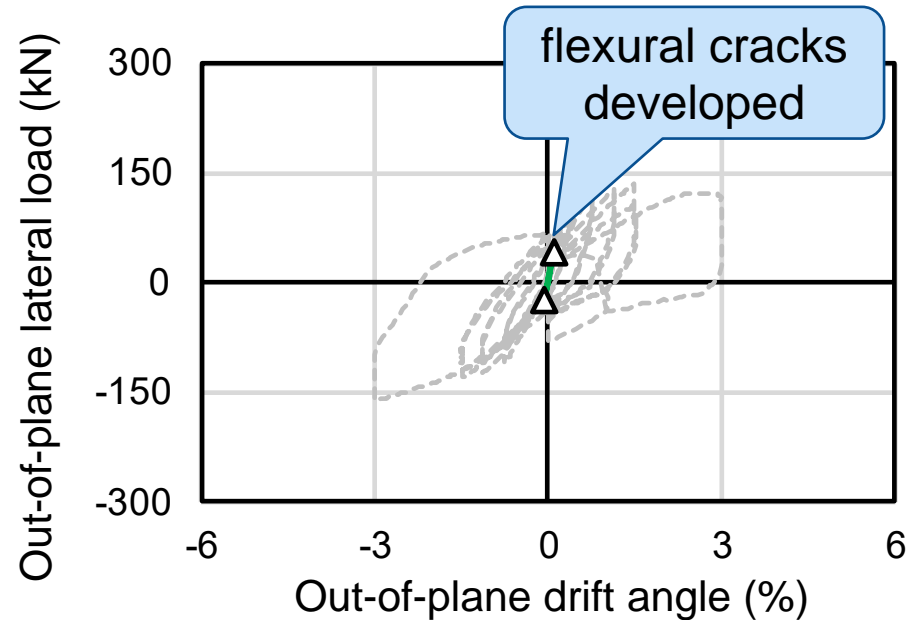
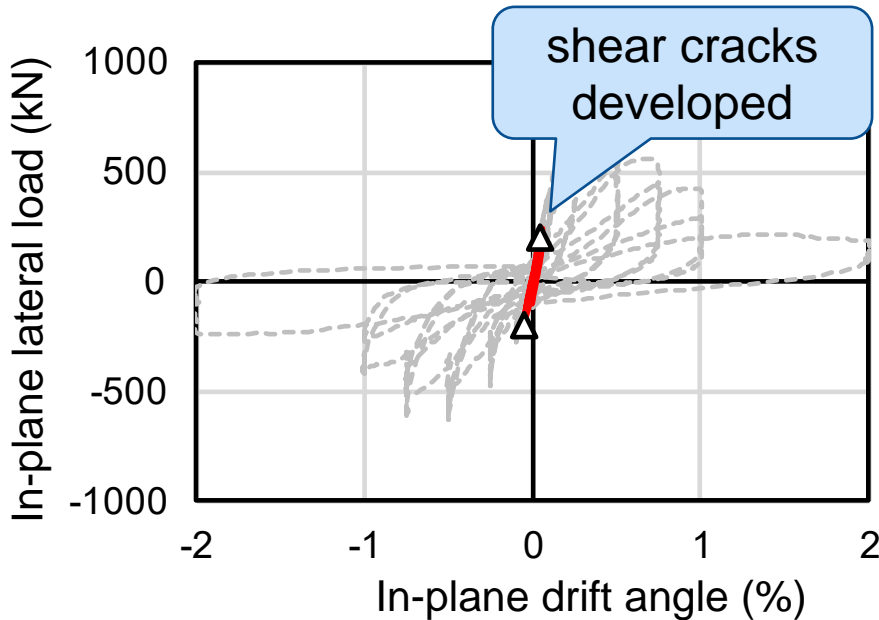


# Experimental Results

WB15-C20T33 O/l drift ratio: 1.5, Axial load:  $-0.33a_g f_y \sim 0.20A_g f'_c$

until  $R_x=0.05\%$

- Shear cracks were observed on the wall panel.
- Flexural cracks were observed on the side columns.

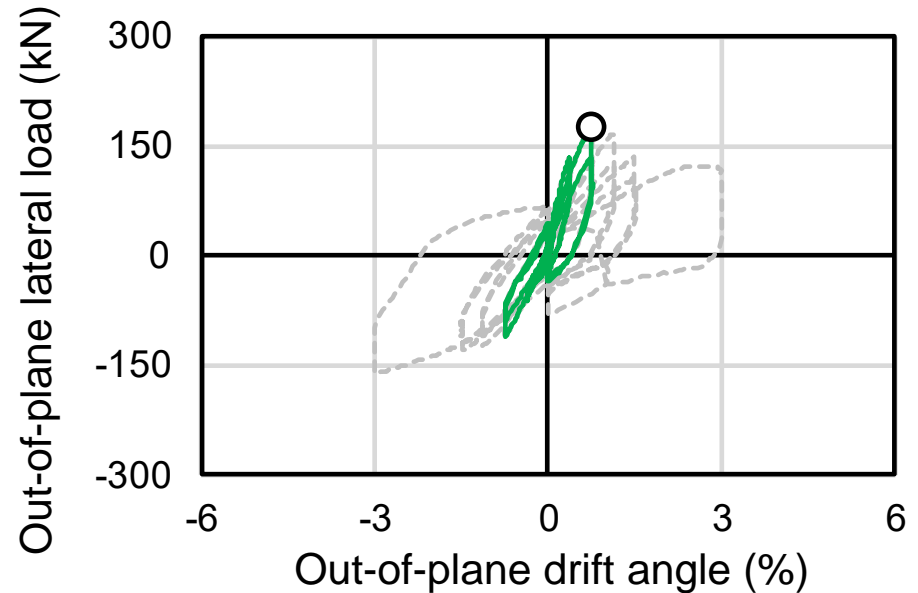
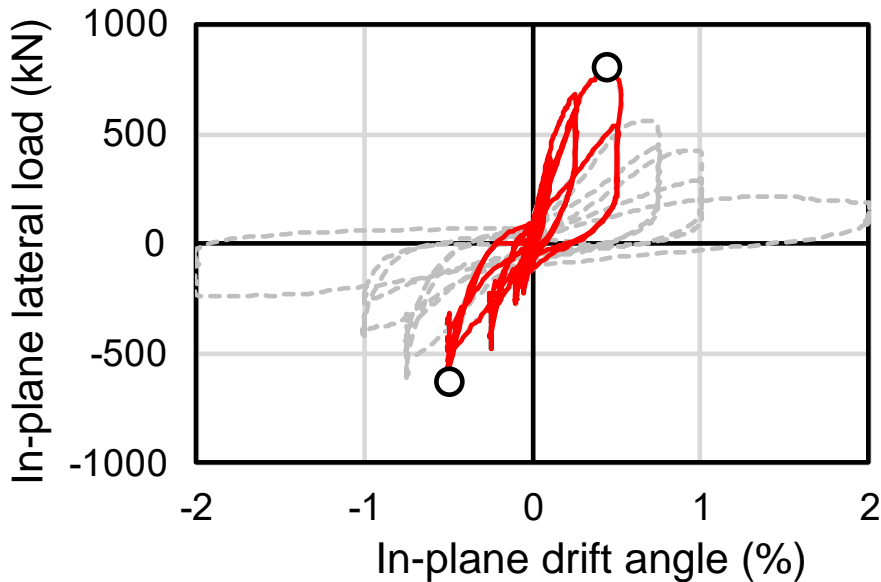
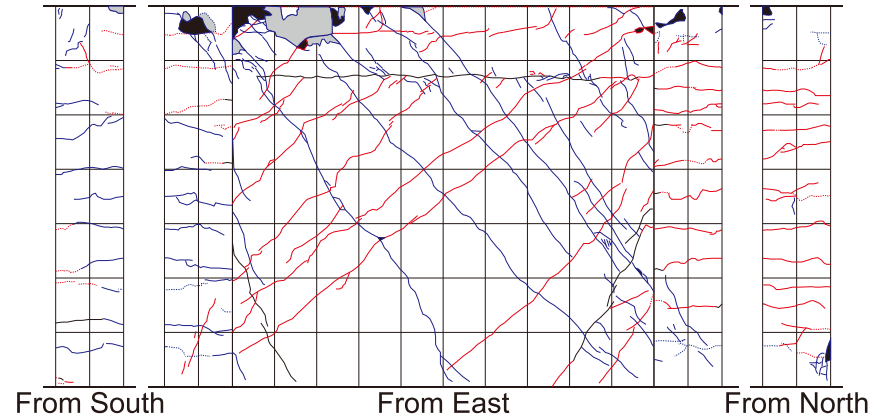


# Experimental Results

WB15-C20T33 O/l drift ratio: 1.5, Axial load:  $-0.33a_g f_y \sim 0.20A_g f'_c$

until  $R_x=0.50\%$

- Shear cracks appeared on the side columns, and maximum load capacity was observed.

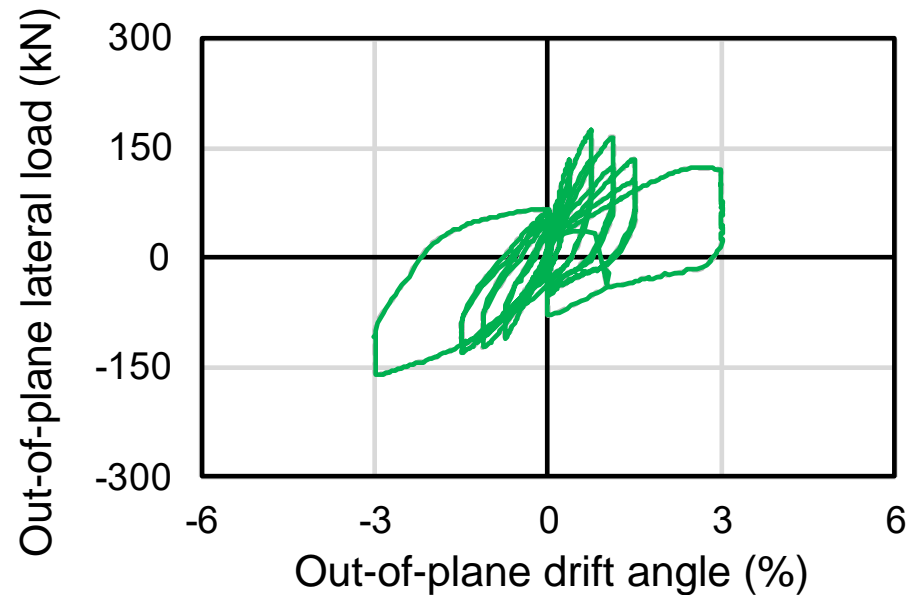
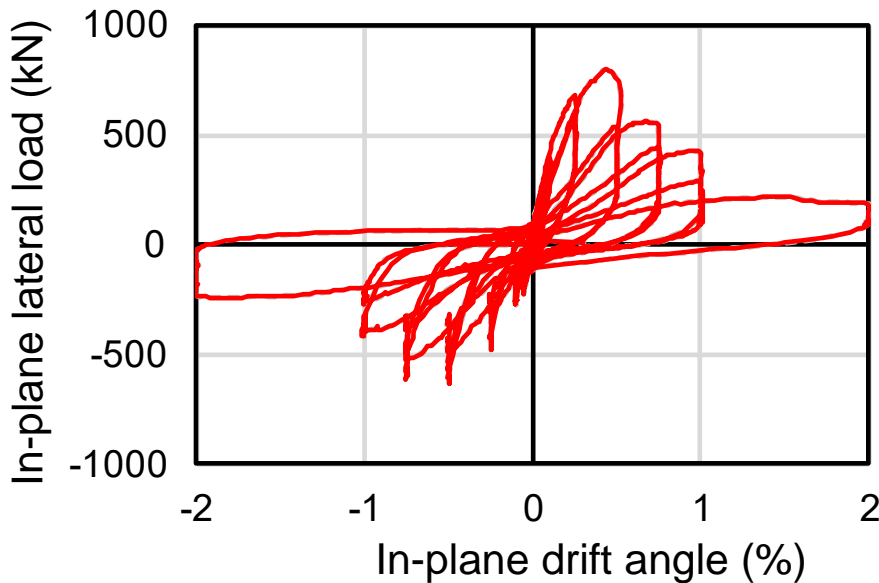
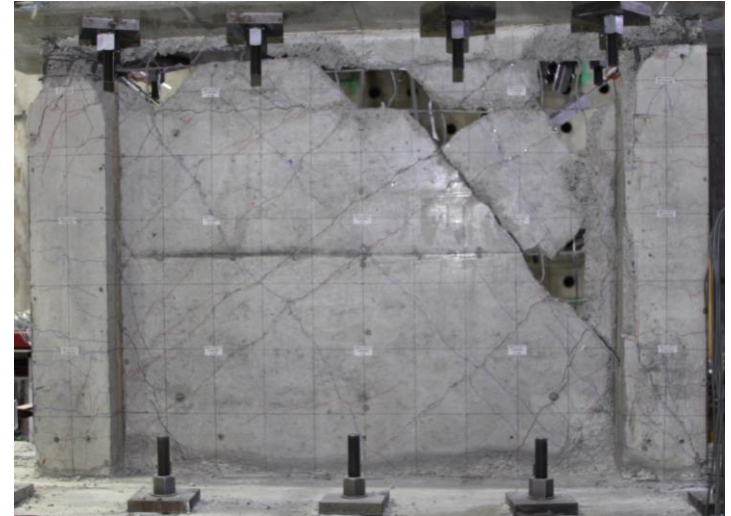


# Experimental Results

WB15-C20T33 O/I drift ratio: 1.5, Axial load:  $-0.33a_{cg}f_y \sim 0.20A_gf'_c$

after test

- The loading was ceased after completion of 1<sup>st</sup> cycle for  $R_x=2.0\%$ .

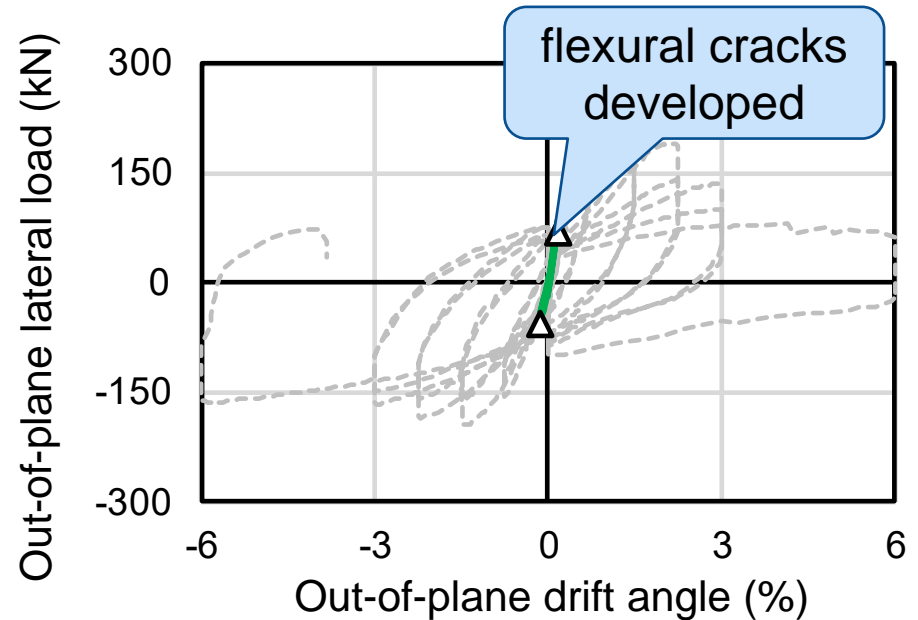
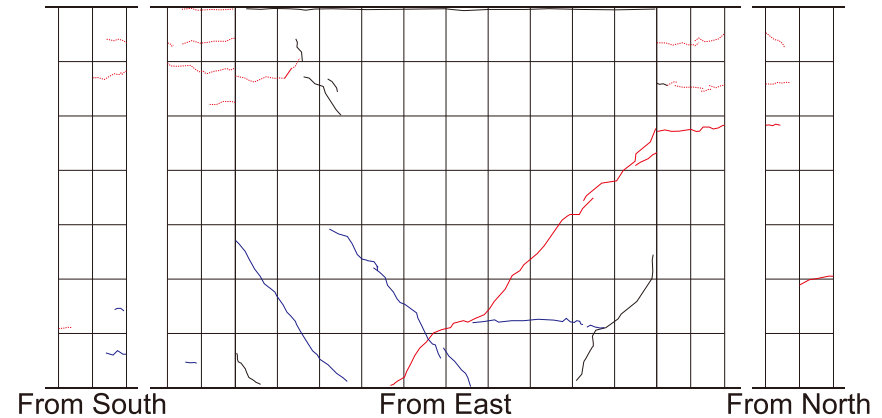
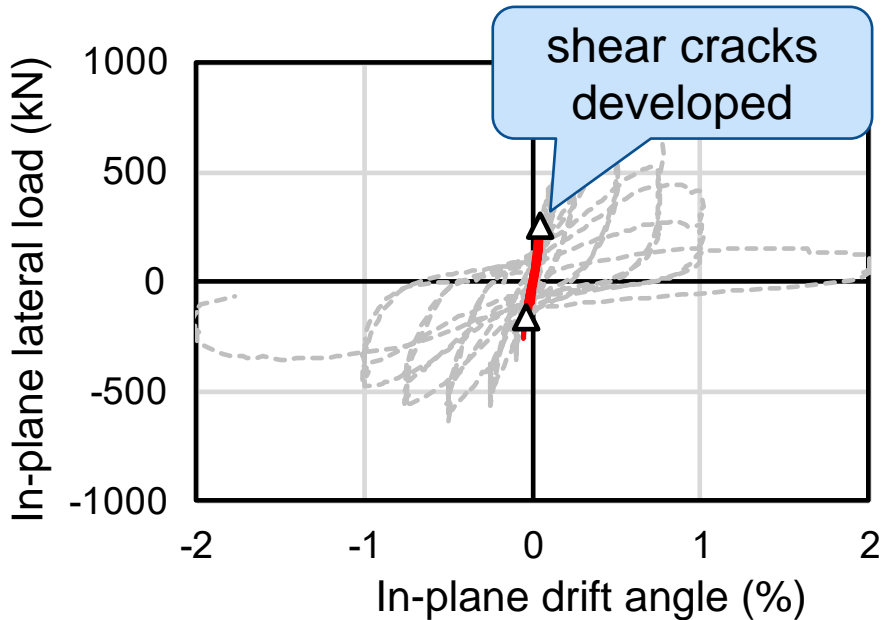


# Experimental Results

WB30-C20T00 O/l drift ratio: 3.0, Axial load:  $0 \sim 0.20A_g f'_c$

until  $R_x=0.05\%$

- Shear cracks were observed on the wall panel.
- Flexural cracks were observed on the side columns.

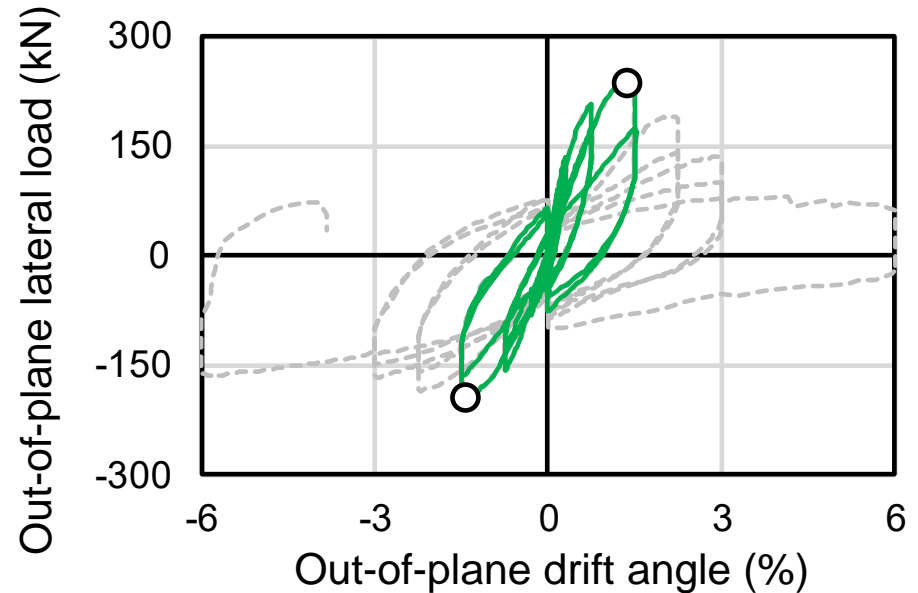
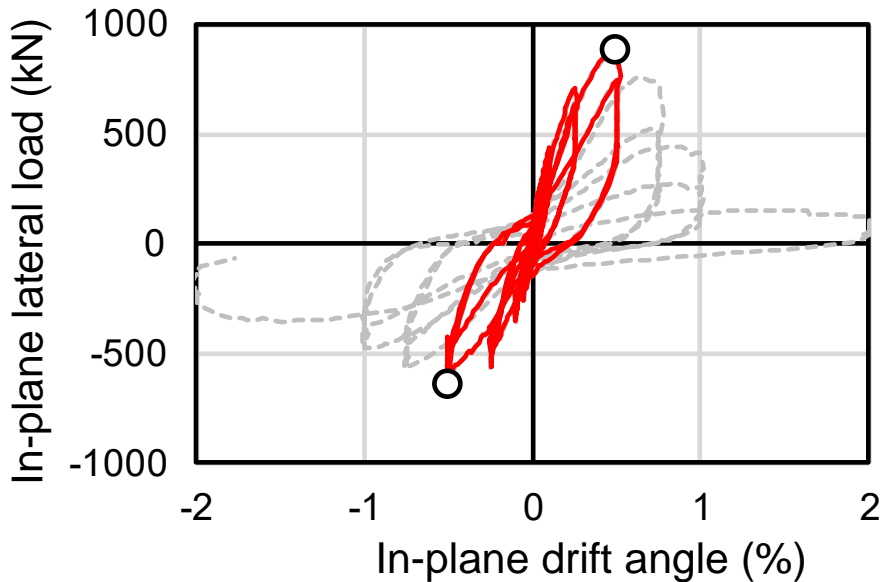
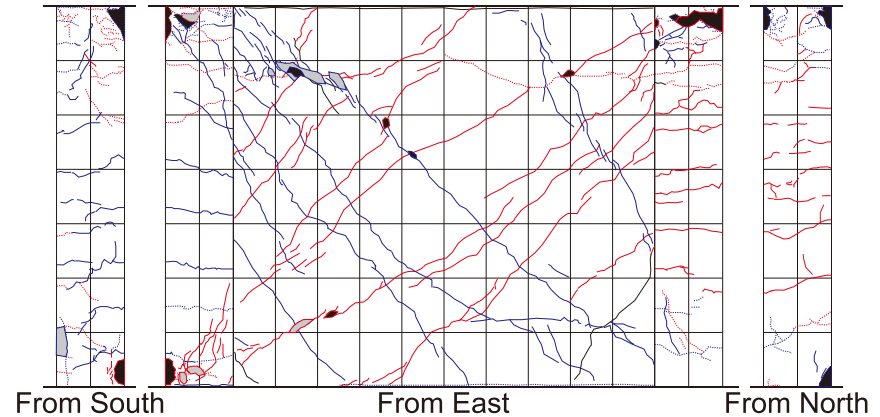


# Experimental Results

WB30-C20T00 O/l drift ratio: 3.0, Axial load:  $0 \sim 0.20A_g f'_c$

until  $R_x=0.50\%$

- Shear cracks appeared on the side columns, and maximum load capacity was observed.

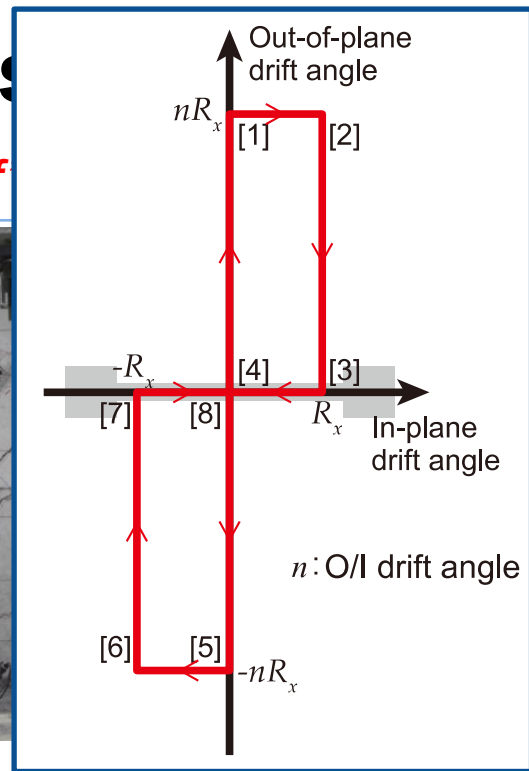
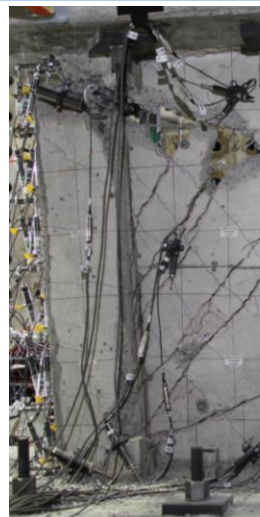
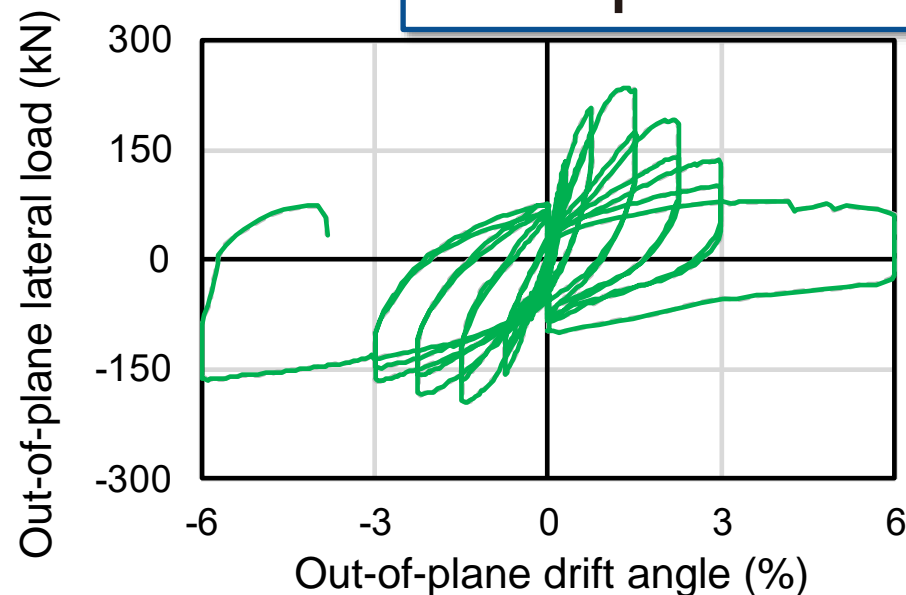
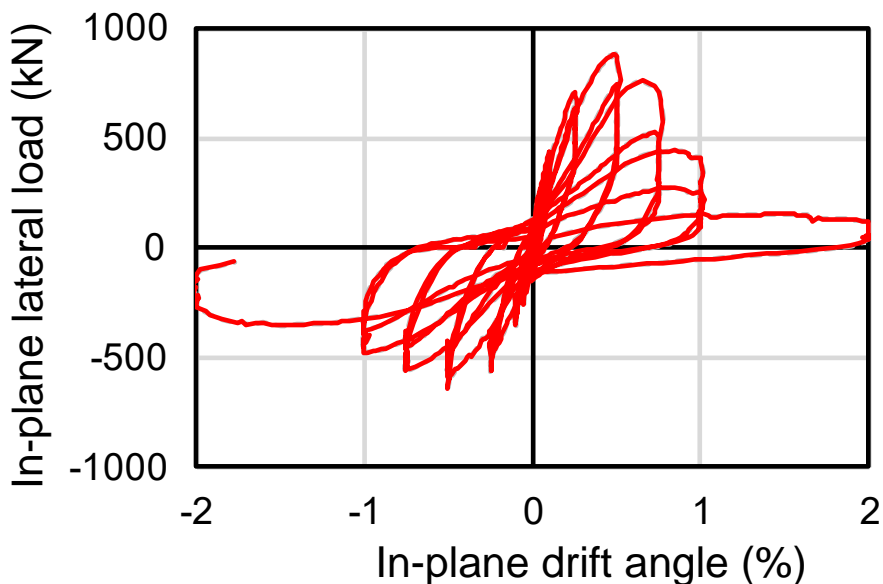


# Experimental Results

WB30-C20T00 O/I drift ratio: 3.0, Axial load:  $0 \sim 0.20A_g f$

after test

- From the peak point [6] to [7] in the first cycle for  $R_x=2.0\%$ , the loading was stopped because the specimen was not able to sustain the axial load.



# Experimental Results

## Envelope curve and maximum load capacity

- The strengths of the concrete of the 4 specimens differed.  
→ **Standardized value**  $\tau_x/f_t$  was calculated and compared.

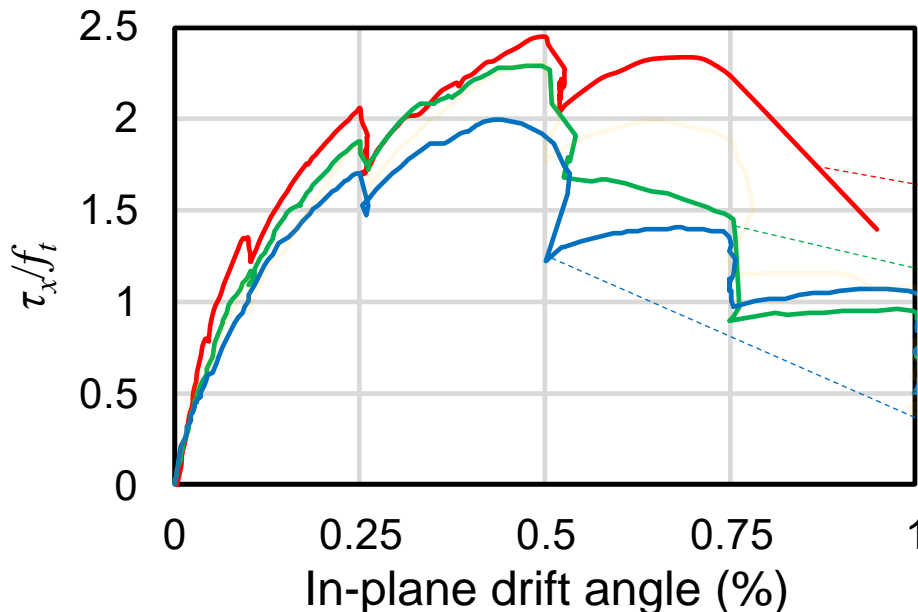
$$f_t = 0.33\sqrt{f'_c}$$

$f'_c$ : compressive strength  
of concrete

$$\tau_x = Q_x / A_g$$

$Q_x$ : in-plane lateral load  
 $A_g$ : gross area of test region

- The larger difference between the maximum and minimum axial loads was, the more significantly the  $\tau_{max}/f_t$  decreased.



Axial load	$\tau_{max}/f_t$ (positive)
$0.20A_g f'_c$ (constant)	2.45
$0 \sim 0.20A_g f'_c$	2.28
$-0.33a_g f_y \sim 0.20A_g f'_c$	2.00

**O/I drift ratio: 1.5**



# Experimental Results

## Envelope curve and maximum load capacity

- The strengths of the concrete of the 4 specimens differed.  
→ **Standardized value**  $\tau_x/f_t$  was calculated and compared.

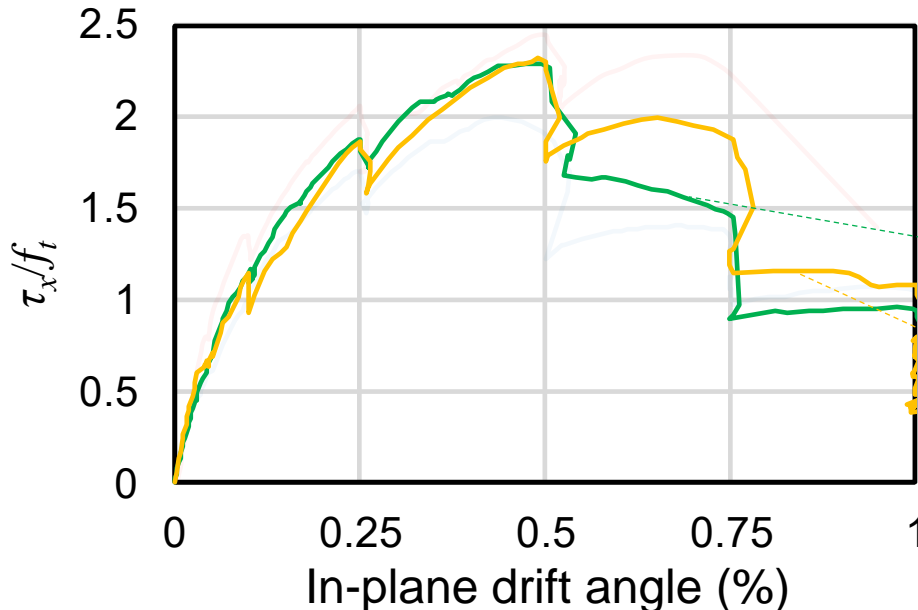
$$f_t = 0.33\sqrt{f'_c}$$

$f'_c$ : compressive strength  
of concrete

$$\tau_x = Q_x / A_g$$

$Q_x$ : in-plane lateral load  
 $A_g$ : gross area of test region

- The out-of-plane deformation had **little influence** on  $\tau_{max}/f_t$ .
- This trend differed from the results obtained by Idosako et al.<sup>1)</sup>.



**O/l drift ratio**       $\tau_{max}/f_t$  (positive)

1.5      2.28

3.0      2.33

**Axial load:** 0 ~ 0.20  $A_g f'_c$

# Experimental Results

## Shear capacity evaluation

- The in-plane maximum load capacity obtained from the experimental results was compared with the calculated shear capacity.

$$Q_{su} = \left\{ \frac{0.068 p_{te}^{0.23} (F_c + 18)}{\sqrt{M / QD + 0.12}} + 0.85 \sqrt{\sigma_{wy} p_{wh}} + 0.1 \sigma_0 \right\} t_e j$$

O/ drift ratio	Axial load		constant				varying					
			$0.20 A_g f'_c$		0 $\sim 0.20 A_g f'_c$				$-0.33 a_{cg} f_y$ $\sim 0.20 A_g f'_c$			
1.5	883kN		1.31		842kN		1.22		800kN		1.07	
	676kN				692kN				750kN			
3.0	888kN		1.24		714kN		experiment		exp./		calc.	
									calculation			

# Experimental Results

## Shear capacity evaluation

- The safety margin decreases as the difference between the maximum and minimum axial loads becomes larger.
- The safety margin was not affected by the out-of-plane deformation.

O/ drift ratio	Axial load		constant		varying			
			$0.20A_g f'_c$		$0 \sim 0.20A_g f'_c$		$-0.33a_{cg} f_y \sim 0.20A_g f'_c$	
1.5	883kN	1.31	841kN	1.22	801kN	1.07		
	676kN		691kN		751kN			
3.0			888kN	1.24			experiment	upper/ lower
			714kN		calculation			

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# Conclusion

- The bigger difference between the maximum axial load and the minimum axial load was, the lower the maximum load capacity the specimen showed.
- Out-of-plane deformation did not affect in-plane maximum load capacity.
- With analytical research, we try to figure out the resistance mechanism of the shear walls under tri-axial loading.

**Thank you for your attention.**